

### Solution to Problem 230A

**Continuity:**

$$\frac{\partial \rho}{\partial t} + \frac{\partial(\rho u)}{\partial x} + \frac{\partial(\rho v)}{\partial y} + \frac{\partial(\rho w)}{\partial z} = 0$$

Since the flow is planar and incompressible this simplifies to:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

Since the velocity,  $v$ , normal to the plate is zero everywhere in the flow it follows from continuity that

$$\frac{\partial u}{\partial x} = 0$$

so  $u$  is only a function of  $y$ ,  $u = u(y)$ .

**Navier-Stokes:  
x-direction:**

$$\rho \left[ \frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z} \right] = -\frac{\partial p}{\partial x} + \mu \left[ \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} + \frac{\partial^2 u}{\partial z^2} \right]$$

Since the flow is planar, since  $v = 0$  and  $\frac{\partial u}{\partial x} = 0$ , and since the pressure is constant, this becomes:

$$\rho \frac{\partial u}{\partial t} = \mu \frac{d^2 u}{dy^2}$$

We use separation of variables to solve this partial differential equation. Assume

$$u(y, t) = Y(y)T(t)$$

Substituting this into the partial differential equation and rearranging, the result can be written as a term which is a function only of  $y$  equal to a term which is a function only of  $t$ . It follows that both must be equal to a simple constant,  $\lambda$ :

$$\frac{1}{T} \frac{dT}{dt} = \frac{\mu}{\rho} \frac{1}{Y} \frac{d^2 Y}{dy^2} = \lambda$$

The equation for  $t$  is then:

$$\frac{dT}{dt} = \lambda T$$

and the solution to this is:

$$T(t) = c_1 e^{\lambda t}$$

The equation for  $y$  is:

$$\frac{d^2 Y}{dy^2} - \frac{\rho}{\mu} Y = 0$$

and the solution to this is:

$$Y(y) = c_2 e^{\sqrt{\rho\lambda/\mu} y} + c_3 e^{-\sqrt{\rho\lambda/\mu} y}$$

The boundary conditions at the plate and as  $y \rightarrow \infty$  are respectively

$$u(0, t) = U(t) = U e^{kt}$$

and

$$u(y \rightarrow \infty, t) = 0$$

The second condition yields  $c_2 = 0$ . It follows that the solution for  $u(y, t)$  is:

$$u(y, t) = c_4 e^{-\sqrt{\rho\lambda/\mu} y} e^{\lambda t}$$

where  $c_4 = c_1 c_3$ . Applying the no-slip boundary condition at the surface of the plate:

$$u(0, t) = c_4 e^{\lambda t} = U e^{kt}$$

so the values of the unknown constant  $c_4 = U$  and  $\lambda = k$  are now determined. This yields a velocity profile:

$$u(y, t) = U e^{kt} e^{-\sqrt{k/\nu} y}$$

where  $\nu$  is the kinematic viscosity  $\nu = \mu/\rho$ . The vorticity,  $\omega(y, t)$ , is given by

$$\omega(\mathbf{y}, \mathbf{t}) = \nabla \times \mathbf{u} = -\frac{\partial u}{\partial y}$$

$$\omega = U \sqrt{\frac{k}{\nu}} e^{kt} e^{-\sqrt{\rho k/\mu} y}$$

The boundary layer thickness,  $\delta$ , is defined as that distance from the plate where the velocity is 10% of the plate velocity:

$$\begin{aligned} 0.1 U e^{kt} &= U e^{kt} e^{-\sqrt{k/\nu} \delta} \\ 0.1 &= e^{-\sqrt{k/\nu} \delta} \\ \delta &= \ln(10) \sqrt{\frac{\nu}{k}} \end{aligned}$$